# Multiphase-Tolerant Turbine (MuTT) Development for an sCO<sub>2</sub> PTES System

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#### **Project Partners:**

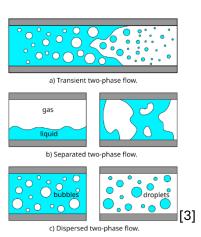
Echogen & Flowserve

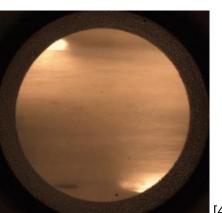




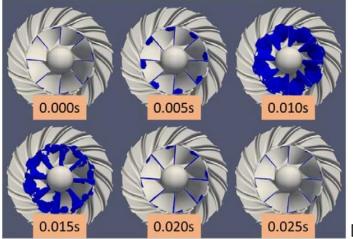
## Why is a cold turbine difficult?

- Multiphase flow!
- Lack of validation data for loss models in this flow regime [3-5]
- Lack of turbine design experience in the open literature [6-7]

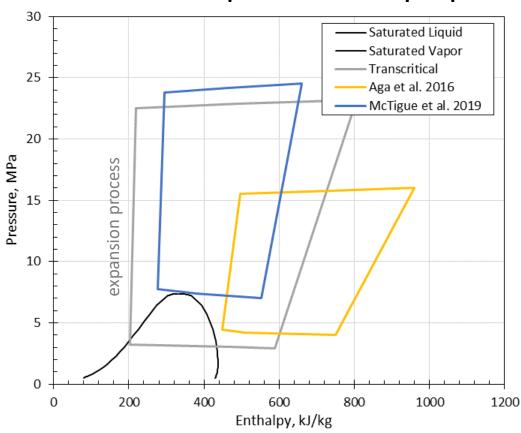








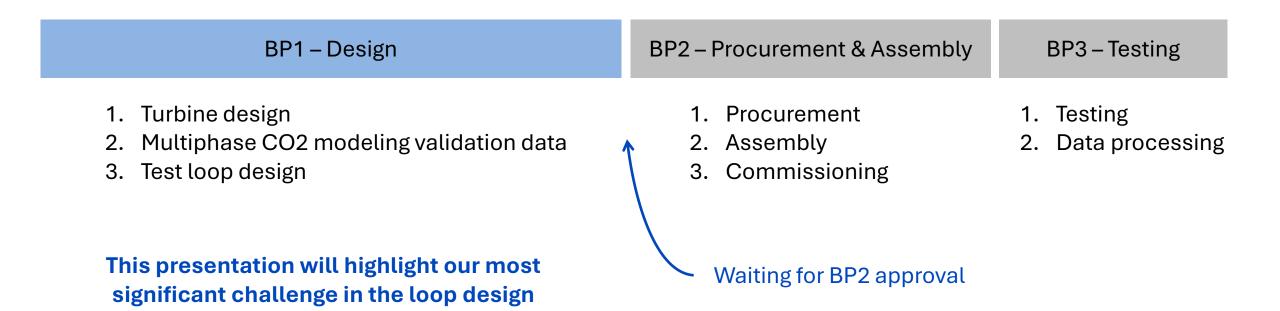
#### Various examples of sCO2 heat pumps



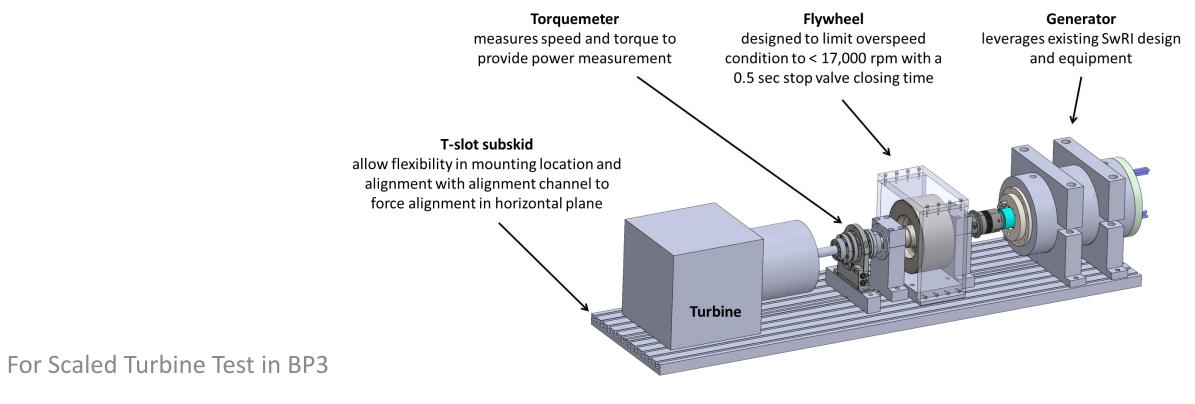
## Multiphase Tolerant Turbine (MuTT) Project Overview

#### **Project Objective:**

Retire multiphase turbine design risks through an understanding of multiphase turbine operation, enabling improved turbine performance and operability range



& initial multiphase data



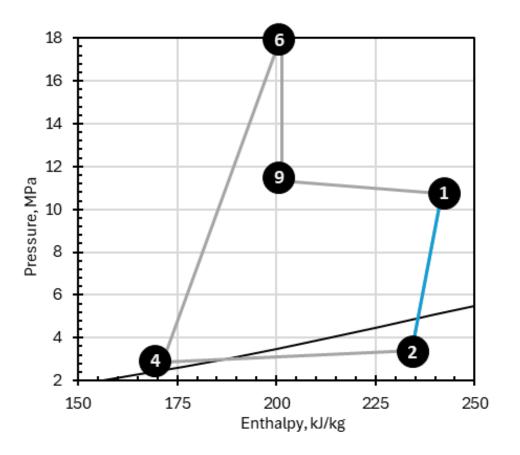
## **MUTT LOOP DESIGN**

Turbine Test Loop Arrangement Existing liquid CO2 pump Throttle control upstream of heater for pressure rating Flow Meter Pump Chiller 2 Multiple coolers Recycle and chillers Hot Density measurement Water at turbine inlet and exit Loop Chiller 1 TPD TPDScaled Existing water-CO2 prototype test heat exchanger **Turbine** turbine

## **Test Loop Conditions Definition**

Two key challenges arose after initial sweeps of test conditions:

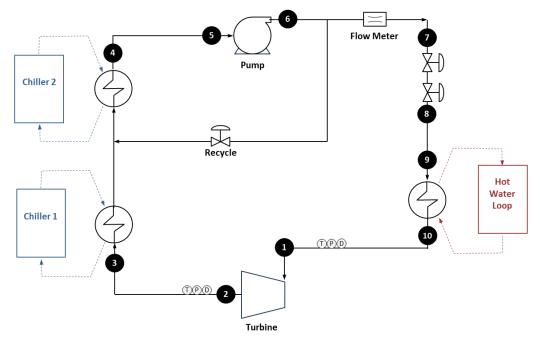
- 1) confidence in pressure loss modeling
- 2) the large required thermal duty for heat rejection

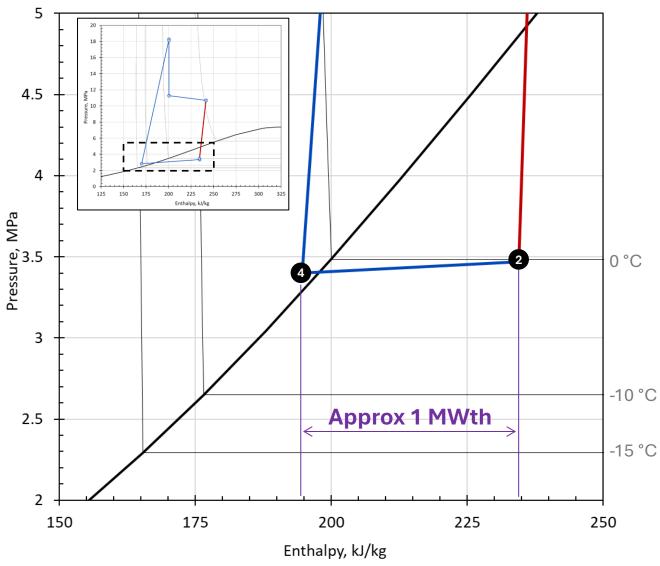


pressure losses...

#### **Assume:**

- 1. Neglect pressure loss in the piping
- 2. Minimal pressure loss through the HX
- 3. No subcooling

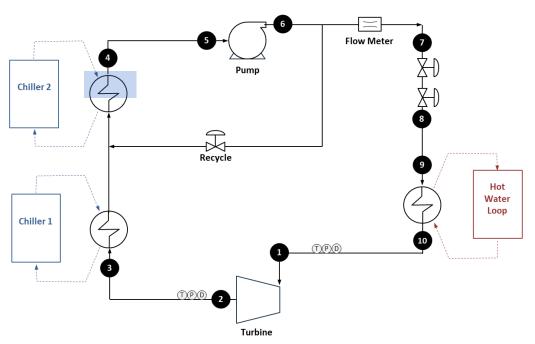


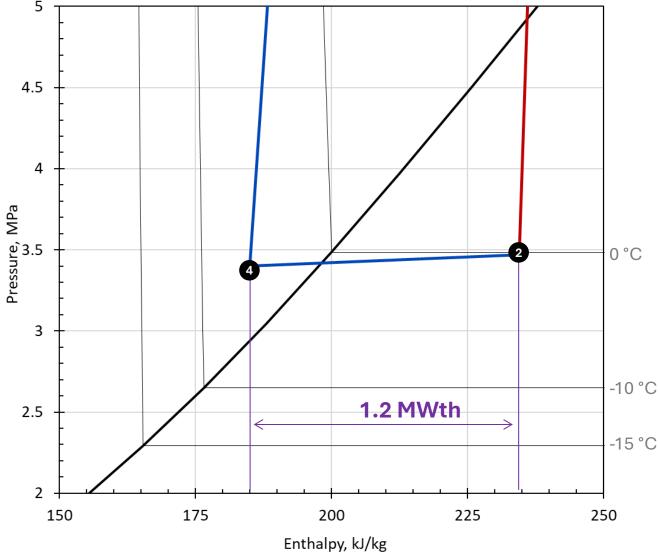


pressure losses...

#### **Assume:**

- 1. Neglect pressure loss in the piping
- 2. Minimal pressure loss through the HX
- 3. No subcooling 5 °C subcooling

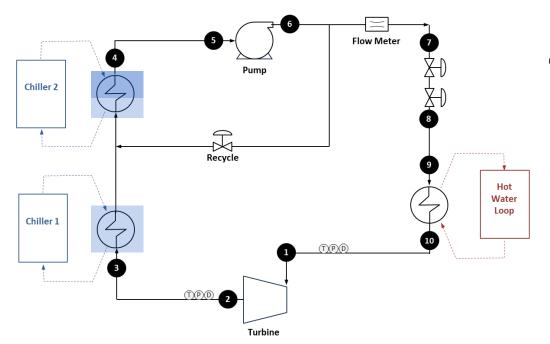


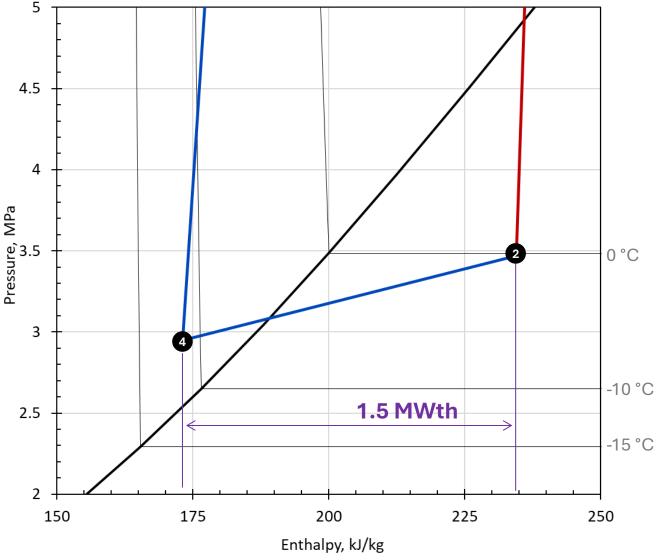


pressure losses...

#### **Assume:**

- 1. Neglect pressure loss in the piping
- 2. Minimal 15% pressure loss through the HX
- 3. No subcooling 5 °C subcooling





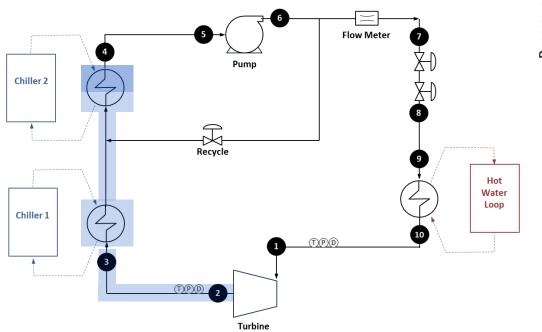
pressure losses...

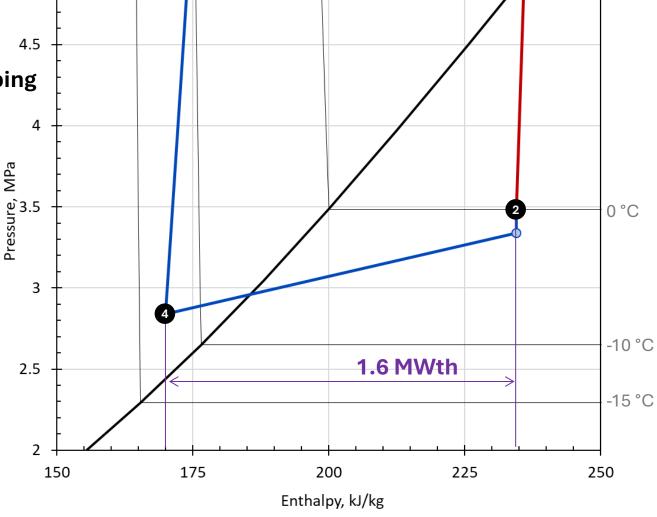


1. Neglect Multiphase pressure loss in the piping

2. Minimal 15% pressure loss through the HX

3. No subcooling 5 °C subcooling





Assumptions of pressure losses underestimated the required

cooling duty by 0.6 MWth

#### **Assume:**

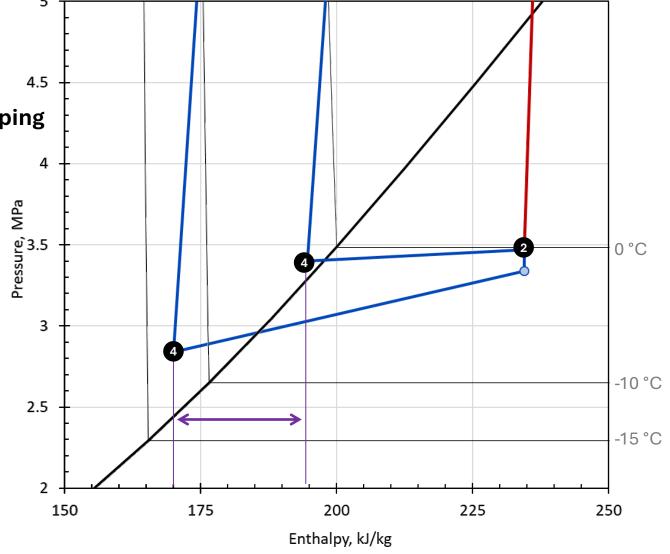
1. Neglect Multiphase pressure loss in the piping

2. Minimal 15% pressure loss through the HX

3. No subcooling 5 °C subcooling

These are all assumptions easily made during proposal and rough order of magnitude checks and led to a large miss in test requirements.

Highlighted the need for multiphase  ${\rm CO}_2$  validation data for not only the turbine design, but also the loop design

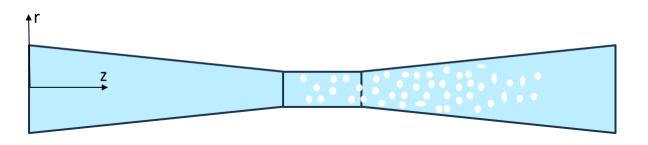




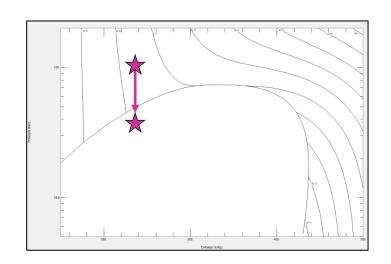


## **MULTIPHASE NOZZLE TESTING**

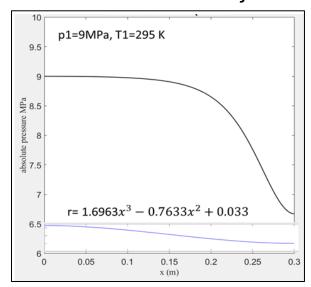
# A converging-diverging nozzle design selected pressure loss and multiphase CFD validation data



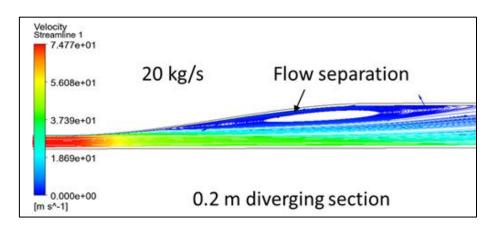
 Liquid at inlet and expand to the highest vapor fraction possible



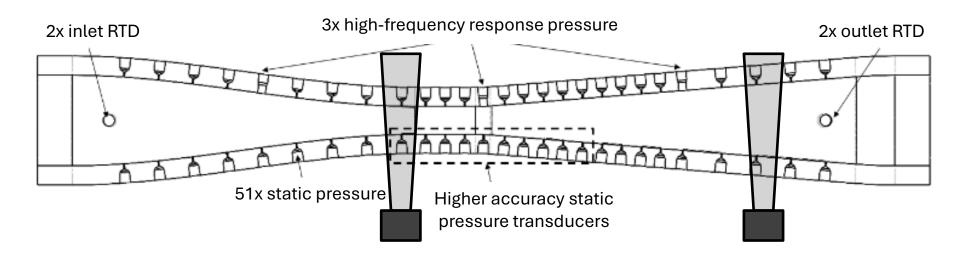
2. Inlet design to accommodate measurability and manufacturability



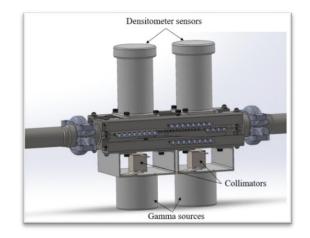
3. Throat was made as long as possible, and the diverging section was made linear and long enough to prevent flow separation

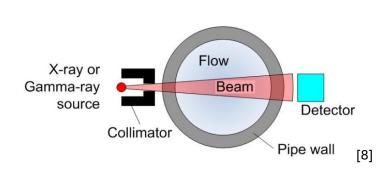


#### Detailed nozzle instrumentation included:

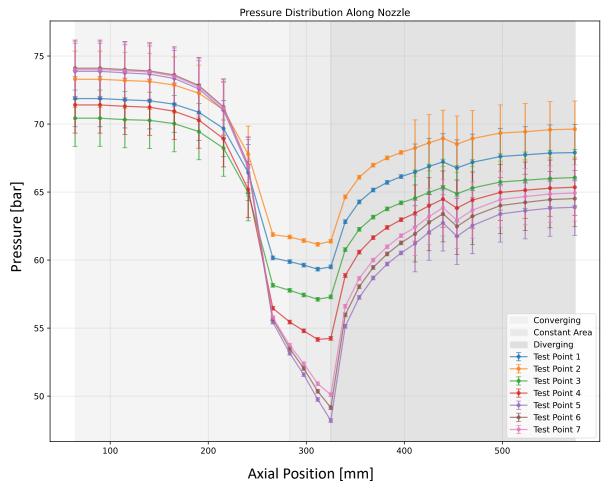


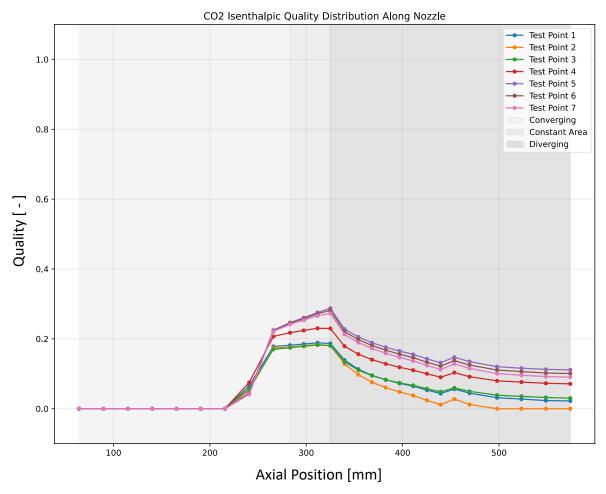
2x Gamma-ray densitometers





## Multiphase flow was confirmed through an isenthalpic process assumption and densitometer data

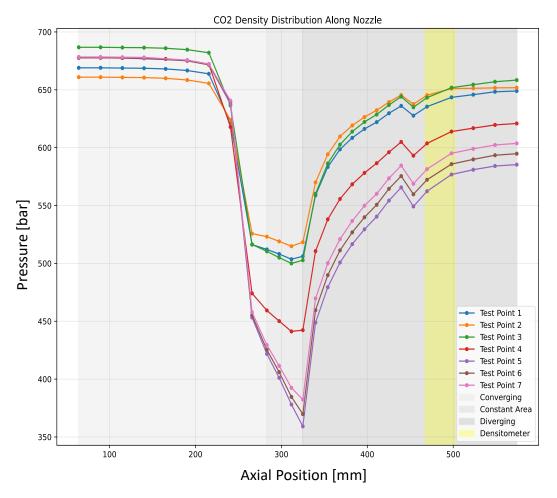


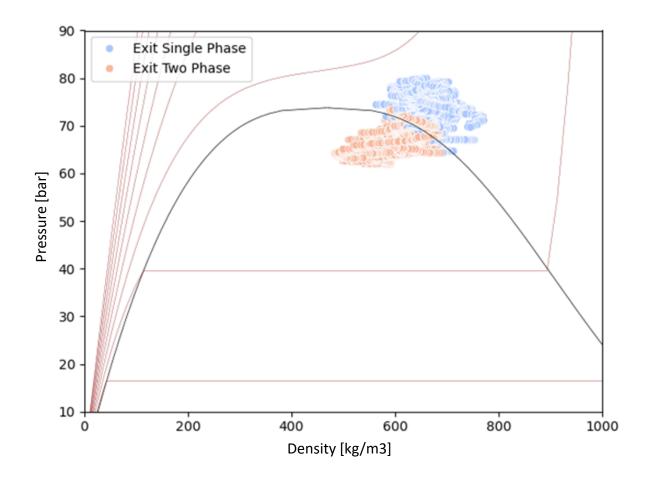


Pressure data showed pressure drops between 4.0-9.6 bar

**Quality estimated from isenthalpic process** 

## Multiphase flow was confirmed through an isenthalpic process assumption and densitometer data

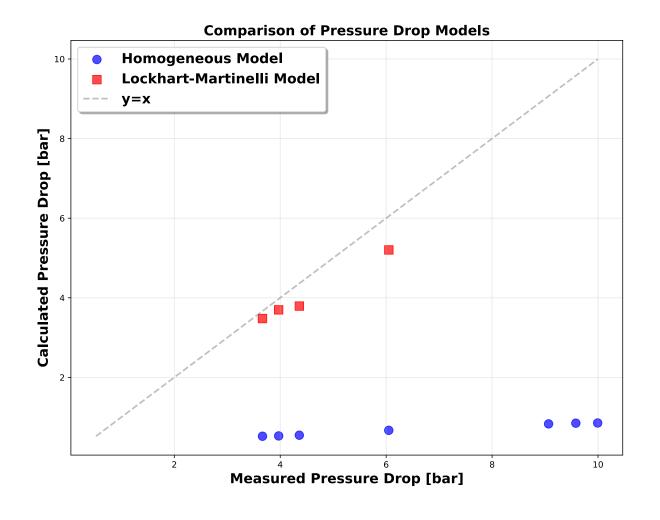




**Density estimated from isenthalpic process** 

Density measurements from densitometer

# The Lockhart-Martinelli multiphase pressure drop model is much more appropriate than homogenous Darcy-Weisbach



#### Homogenous Model: Darcy-Weisbach [9]

$$\Delta P_{total} = \int_0^L f(x) \frac{dx}{D(x)} \frac{\rho(x)v(x)^2}{2}$$

$$f = \text{Darcy friction factor}$$

$$L = \text{pipe length}$$

$$D = \text{pipe diameter}$$

$$\rho = \text{fluid mixture density}$$

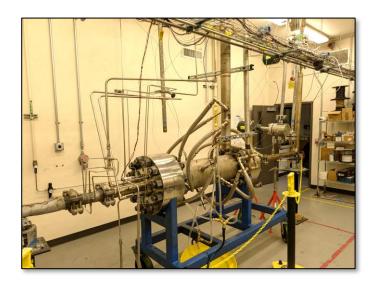
$$v = \text{fluid mixture velocity}$$

## Lockhart-Martinelli Model [10]: Two-phase pressure drop multiplier

$$\phi_{lo}^2 = 1 + \frac{20}{X} + \frac{1}{X^2}$$

$$X = \sqrt{\left(\frac{f}{2D}\right) \cdot \frac{(1 - VMF)^2}{VMF^2} \cdot \left(\frac{\rho_v}{\rho_l}\right)}$$







STEP, Oxy-combustion, Compander with IGV

### OTHER SCO2 TESTING ACCOMPLISHMENTS

## STEP sCO<sub>2</sub> Technology Maturation Achievements



- Successfully demonstrated gas-fired indirect sCO<sub>2</sub> plant operation at 500 °C simple recuperated cycle "max" conditions generating ~4 MW net power while grid-synchronized
- All major components commercially procured except turbine jointly designed by GE Vernova and SwRI:
  - Compressors: Baker Hughes
  - Heat Exchangers: Parker Heatric, Optimus, Vacuum Process Engineering
  - Heater Protection Valve and 500 °C Turbine Trip Valve: SchuF, AVS/HORA
  - Plant Controller: GE Mark VI
- Demonstrated repeatability through multiple operations, also safely demonstrated fast and slow trips
- **Project Partners:**











Plant design details available to JIP members

























### Thank you

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The Design of a Multiphase CO2 Turbine Test Facility

GT2025-154157

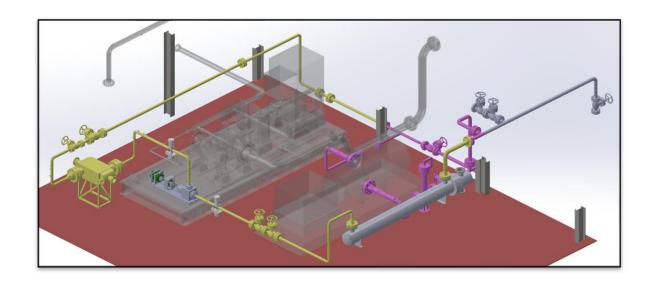
**Experimental Investigation of Vapor Formation in Liquid CO2 Flow Through a Converging-Diverging Nozzle** 











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